## Research project and supervisor team

| Supervisory          | Dr. Kean-How Cheah  |
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| Team                 |   |
|                      | Dr. Yong Ren  |
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|                      | Dr. ChungKet Thein  |
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| Short introduction & |   |
| description of       | Nanosatellites $(< 10 \text{ kg})$ have gained much attention among the space community |
| research project     | Nanosatemites (< 10 kg) have gamed much attention among the space community             |
| research project     | as a rapid and cost-effective platform to demonstrate new technologies in outer         |
|                      | space. However, most of the nanosatellites have no onboard propulsion system            |
|                      | due to the limited spatial volume and electrical power. The need of propulsive          |
|                      | capability for the nanosatellites to perform more complex mission has prompted          |
|                      | an intensive research and development in the area of micropropulsion systems.           |
|                      | Using microelectromechanical system (MEMS) technology, the existing chemical            |
|                      | propulsion systems have been miniaturized significantly. Unfortunately, the high        |
|                      | thermal conductivity of silicon has resulted in a low system efficiency as a result     |
|                      | of onbanced best loss at micro cools. In addition, the further integration of liquid    |
|                      | or enhanced near loss at micro-scale. In addition, the further integration of liquid    |
|                      | chemical micropropulsion system into nanosatellite is limited by the heavy and          |
|                      | bulky air pressurized fluid handling system. This project aims to address these         |
|                      | issues by studying the solid sublimation at reduced pressure and subsequently           |
|                      | implement it into the development of a space micropropulsion system. Additive           |
|                      | manufacturing technology (3D printing) will be used to fabricate a ceramic based        |
|                      | prototype with an axisymmetric conical micronozzle. Performance of the                  |
|                      | prototype with an axisymmetric conical micromozzie. Terrormance of the                  |
|                      | prototype will be evaluated through a series of experiments using an in-house           |
|                      | built torsional micronewton thrust stand. Upon completion of the project, a new         |
|                      | and highly compact micropropulsion system which uses green solid propellant             |
|                      | will be demonstrated.   |
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| Contact points       | Dr. Kean-How Cheah  |
|                      | Email: Kean-bow cheah@nottingham edu cn   |
|                      | Lindi. <u>Rear now.circan@nottingnam.cdu.cn</u>   |
|                      |   |